

SSC DETECTOR SOLENOID DESIGN NOTE #115

TITLE: External Pressure Design and Layout of Chimney for the SDC Solenoid

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DATE: October 15, 1990

ABSTRACT: This Design Note contains the design standards and specifications for the chimney, preliminary design calculations for a rectangular chimney, design calculations for a cylindrical chimney and

oraliminary chimney dimensions for a cylindrical chimney and

preliminary chimney dimensions.

external pressure.

SUMMARY: The design calculations for both the square and the cylindrical chimney were done in accordance with Section VIII, Division 1 of the ASME Pressure Vessel Code. Design rules for the square chimney are in Appendix 13, Vessels of Noncircular Cross Section. Design rules for the circular chimney are in Part UG. ASME/ANSI B31.3, Chemical Plant and Petroleum Refinery Piping, specifies that Section VIII, Division 1 of the ASME Pressure Vessel Code be followed to determine wall thickness and stiffening requirements for straight pipe under

The calculation for a cylindrical chimney shows that the chimney can be fabricated with 12ⁿ Schedule 10 aluminum pipe. The wall thickness is 0.18ⁿ. The outside diameter is 12.75ⁿ. The maximum allowable external pressure is 25.4 psi. The collapse pressure safety factor is 3.25. Circumferential stress at the collapse pressure is 890 psi.

The calculation for a square, unstayed, aluminum chimney indicates a minimum required wall thickness of 0.875° to 1° . This calculation is preliminary because I have two unresolved questions about the design rules in paragraph 13-14 of Appendix 13. The questions are discussed on page 9.

A square chimney stayed at midlength may have thinner walls. No calculations were performed on this type of chimney.

CONCLUSION:

At this time, a cylindrical chimney is preferred over a square one because the cylindrical design has a much smaller radiation thickness. There appears to be sufficient room in it for all the required cryogenic lines, an MLI insulated shield, the coil leads and instrumentation wiring.

The relieving capacity and vacuum conductance of this cylindrical chimney must be checked before it is chosen over the square chimney.



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SUBJECT

EXTERNAL PRESSURE DESIGN AND LAYOUT OF CHIMNEY FOR THE SDC SOLENOID

NAME A. M. STEFANIK
DATE 8/30/90 REVISION DO REVISION DATE

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DESIGN SPECIFICATIONS AND STANDARDS

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- 1. MINIMIZE RADIATION THICKNESS => ALUMINUM CONSTRUCTION
- 2. ASME SEC VIII, DIV 1, AS DIRECTED BY ANSI B31.3.
- 3. MAXIMUM OUTSIDE DIMENSIONS: 12" SQUARE
- 4. CHIMNEY CONNECTS TO CRYOSTAT AT TOP (0°) AND RUNS STRAIGHT UP THROUGH THE DETECTOR.



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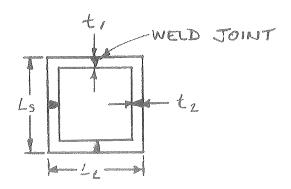
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SQUARE OUTER WALL

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DESIGN IN ACCORDANCE WITH ASME SECTION VIII DIV 1 APP. 13
GEOMETRY #1 LL=Ls=12"; t_=ts; No STAYS



CHIMNEY LENGTH = Lv = 35'

CHIMNEY WIDTH = H= 12"

CHIMNEY DEPTH = h = 12"

ASPECT RATIO = LV/H = 35'/1' = 35 > 4 (13-4(h))

MATERIAL: 5083-0 ALUMINUM

TABULATE ALLOWABLE STRESSES (APP 13, PARA 13-4)

J = 9,800 psi

Sman S = 9,800 psi IN BASE METAL

Sma SE = 9,800 Epsi AT WELD JOINT

THE LONGITUDINAL JOINTS ARE CATEGORY A, TYPE NO. 2 JOINTS. CIRCUMFERENTIAL JOINTS ARE CATEGORY B OR C, TYPE NO. 2. USE FULL RADIOGRAPHIC EXAMINATION.

LE = 0.9 FOR ALL OF THE WELDED JOINTS. NOTE 3 OF THE JOINTS COULD BE TYPE!

Smaw = 9,800 (0.9) = 8,820 psi

STa = 1.55E = 1.5 (9,800) (0.9) = 13,230 psi

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EXTERNAL PRESSURE DESIGN REFR: APP 13, PARA 13-14 (a)(1)

Pe = 15 psí $t_1 = t_2 = t$ Assume t = 3/8'' = 0.375''

Sm = PH/2t = 15psi(12in) 2(in)

 $(S_b)_M = \frac{gh^2c}{t^3} \left[-1.5 + \frac{(1+\alpha^2K)}{1+K} \right]$ SQUARE CHIMNEY BECAUSE THICKNESS REQUIREMENT.

C = t/2 = "/2 = a = H/h = 12"/12"=1 K= (12/1) a =/

 $(5b)_m = 15(12)^2(0.)$ $(0.)^3 \left[-1.5 + (1+(1)(1))\right]$

NOTE: THERE IS NO NEED TO FINISH THIS CALCULATION BECAUSE THE WALL THICKNESS NEEDED FOR STABILITY IS MUCH GREATER THAN THE WALL THICKNESS OF A CYLINDRICAL CHIMNEY, THIS INDICATES THAT A CYLINDRICAL CHIMNEY IS BETTER THAN A

> CALCULATIONS FOR STABILITY OF THE SQUARE CHIMNEY AND FOR THE DESIGN OF THE CYLINDRICAL CHIMNEY ARE ON PAGES 5 THROUGH 10.

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CHECK SIDE PLATE STABILITY REFR.: APP 13 PARA 13-14 (a)(2)

ASSUME 3/8" THICK SIDE PLATES. \Rightarrow $t_1 = t_2 = \frac{3}{8}$ " = 0.375" (SAME AS PG 4) $\frac{4S_{mA}}{S_{crA}} + \frac{4S_{mB}}{S_{crA}} + \frac{S_{bA}}{S_{crA}} + \frac{S_{bB}}{S_{crB}} = \frac{5}{1.5}$

 $SmA = \frac{PehH}{2(t,H+tzh)} = \frac{15(12)(12)}{2(0.375(12)+0.375(12))} = 120$

 $SmB = \frac{Peh}{2t_1} = \frac{15(12)}{2(0.375)} = 240$

 $\frac{L_V}{H} = \frac{35'}{1'} = 35 \Rightarrow J_B = 0.125 + J_A = 0.0375$

 $56A = 6 J_A H^2 Pe = 6 (0.0375) (12)^2 (15) = 3,456$

 $S_{6}B = 6J_BH^2Pe - 6(0.125)(12)^2(15) = 11,520$

SCRA = 12 (1- V2) (t) KA

LY = 35 => KA = 5.5

 $H = \frac{1}{35} = 0.029 \Rightarrow KB = 15$

v = 0.334 E2 = 10.3 × 10 6 psi

 $S_{CPA} = \frac{\pi^2 \left(10.3 \times 10^6\right)}{12 \left(1-0.334^2\right)} \left(\frac{0.375}{12}\right)^2 5.5 = 51,214$

 $S_y = 16,000 \text{ psi}$ $S_y/2 = 8,000 \text{ psi}$ $S_{CTA} = 51,214 + 5y/2 = 8,000$

: ScrA = Sy-Sy 2/4-SCTA

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ScrA = 16,000 - 16,000 2/(4*ScrA) = 16,000 - (6.4 ×107)/ScrA SCRA = 16,000 ScrA - 6.4 ×107 SCRA - 16,000 SCRA + 6.4 ×107 =0 SOLVE USING THE QUADRATIC EQUATION.

SOLVE USING THE QUADRATIC EQUATION. Q=1; b=-16,000; c=6.4×107

 $SCrA = -6 \pm \sqrt{3^2 - 49C} = 16,000 \pm \sqrt{16,000 - 4(1)(6.4 \times 10^{3})}$ 2a 2(1)

ScrA = 3,000 psi

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SCIB = TEZ (t/) 2 KB

 $= \frac{\pi^{2} (10.3 \times 10^{6})}{12 (1-0.334^{2})} \left(\frac{0.375}{35 \times 12}\right)^{2} 15 = 114$

SCTB = 114 < 8,000 = 1/25y : SCTB = 114

SCrA = 51,214

5'cr3 = 114

CHECK THE INTERACTION EQUITION:

 $\frac{4(120)}{8,000} + \frac{4(240)}{114} + \frac{3,456}{120}$ [1-2+120 1.5(9,800)

 $+ \frac{11,520}{[-2*\frac{240}{14}]} = 0.06 + [8.42] + 0.23 + (-0.244)$

THE TWO CIRCLED TERMS CONTAIN SMB, SCTB, S68 4 S'CrB. FOR A GIVEN LV , ONLY t, CAN BÉ VARIED TO CHANGE THE VALUES OF SMB, SCIB, SLB AND S'CIB. NEGLECT THE "A" TERMS AND SOLVE FOR THE VALUE OF t, THAT BRINGS THE INTERACTION EQUATION TO UNITY.

SmB = Peh = 15(12) = 90 2t, 2t, +.

 $S_{6B} = \frac{6J_BH^2Pe}{t_1^2} = \frac{6(0.125)(12)^2I_5}{t_1^2} = \frac{1620}{t_1^2}$

ScrB = ScrB' = $\frac{\pi^2 E}{12(1-v^2)} \left(\frac{t_1^2}{Lv}\right) K_8 = \frac{\pi^2 (10.3 \times 10^6)}{12(1-0.334^2)} \frac{t_1^2}{(35 \times 12)^2} 15$ = 810.8 t,2 (ONLY TRUE FOR SCIBIF

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$$\frac{4 \text{ SmB}}{\text{ScrB}} + \frac{5bB}{\left(1 - 2 \frac{5mB}{5 \text{ crB}}\right)} \leq 1.0$$

$$\frac{4(90)}{(t_1)} + \frac{1620}{t_1^2} < 1.0$$

$$810.8t_1^2 + \left(\frac{90}{t_1}\right) = 1.5(9,800)$$

$$810.8t_1^2$$

FOR t, = 0.875, INTERACTION EQUATION = 0.88 WITH "A" TERMS.
FOR t, = 0.75, INTERACTION EQUATION = 1.47 INCLUDED)

conclusions:

- 1) WALL THICKNESS FOR THIS CASE WOULD BE ~ 78".
 THE DESIGN OF AN UNSTAYED, SQUARE CHIMNEY WILL NOT
 BE UNDERTAKEN IF A CYLINDRICAL CHIMNEY HAS A
 WALL THICKNESS LESS THAN 7/8".
 - 2) TRY A CYLINDRICAL CHIMNEY.
- 3) TRY A SQUARE CHIMNEY STAYED AT MIDLENGTH IF A CYLINDRICAL CHIMNEY IS TOO THICK OR IF THE INSULATING VACUUM FOR THE LHE & LNZ LINES IS ISOLATED FROM THE PRESSURE RELIEF CHANNEL.

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Note on the calculations for the square chimney

The calculations for the square chimney are preliminary because I have two unresolved questions about the design rules in paragraph 13-14 of Appendix 13.

First, equation (6A) is used to calculate the critical buckling stress, ScrA, for a simply-supported plate in uniaxial, uniform compression along its narrow dimension (about 12^n in this case) when ScrA is less than Sy/2. Equation (6A) yielded 51,214 psi, which is greater than the Sy/2 value of 8,000 psi. Equation (6B) must be used when (6A) is not applicable, that is, for ScrA greater than Sy/2. Solving equation (6B) with the quadratic equation yielded only one value for ScrA, Sy/2. This seems inconsistent because the equation is defined for use when ScrA is greater than Sy/2. Since this equation yields only one value (assuming I have solved it properly), is it correct? The design of the square chimney is based on ScrA equal to Sy/2.

The second question concerns calculating the critical buckling stress, ScrB, for a simply-supported plate in uniaxial, uniform compression along its long dimension (about 420° in this case). The equations for ScrB are the same equations used to solve for ScrA. I calculated a value of 114 psi for ScrB. This low value occurs because the plate width, 420" in this case, is a squared term in the denominator. The calculated values of ScrA and ScrB indicate that the critical buckling stress for a 12" wide x 420" long plate is much greater than for a 420° wide x 12° long plate. Is this correct for the plates in this type of vessel? Reference 1 notes that buckling strength of a plate is found to be independent of length when the ratio of plate length to width is greater than 5. Paragraph 13-4 (h) states that the design equations in Appendix 13 are based on vessels in which the length to side dimension ratio (aspect ratio) is greater than 4. Also, these design equations are conservatively applicable to vessels of aspect ratio less than 4. Is it too conservative to calculate critical buckling stress ScrB for the side walls of this long, square vessel with the same formula used to calculate ScrA?

I have given these questions regarding the ScrA/ScrB design equations to Chuck Grozis. He will look for answers at his ASME pressure vessel class.

Derivation of the critical buckling stress equation used in Appendix 13 of the ASME Code can be found in reference 2.



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The 1989 edition of Roark's Formulas for Stress and Strain book contains formulas for elastic stability of plates and shells in Table 35 on page 684. Equation 1 is used to calculate the critical buckling stress for a rectangular plate under equal uniform compression on the two opposite narrow edges with several manners of support. For a plate with all edges simply supported or for a plate with all edges clamped, equation 1 yields stresses which are lower than the stress calculated with the Appendix 13 equation. Equation 2 considers uniform loads on all four edges of a rectangular plate. The equation applicable to a rectangular plate with all edges clamped, equation 2b, is noted to be most accurate when the plate is nearly square but the accuracy for other dimensions is not stated. Equation 2a is for a rectangular plate with all edges simply supported. It will be solved if it becomes necessary to use a square chimney.

REFERENCES:

- 1) Alexander Blake, Practical Stress Analysis in Engineering Design, Marcel Dekker, Inc., New York and Basel (1982), pg 87
- 2) C. L. Dym and I.H. Shames, Solid Mechanics: A Variational Approach, McGraw-Hill, Inc., USA (1973), pg 499

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CYLINDRICAL OUTER WALL

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EXTERNAL PRESSURE DESIGN REFR: ASME SEC VIII, DIV 1

SET Do = 12.75"

CHIMNEY LENGTH = 35' = 420" T

ASSUME t = 0.25" (SCH20) Do

MATERIAL: SEAMLESS ALUM PIPE SB-241 6061-T6 WLD

FOLLOW UG-28:

Do/t = 12.75/0.25 = 51 > 10

L/Do = 420/12.75 = 32.94

FROM FIGURE 5-490-28.0 , A = 0.00045

FROM FIGURE 5-UNF-28.30, B= 2,250

 $Pa = \frac{4B}{3(D_0/t)} = \frac{4(2,250)}{3(51)} = 58.8 \text{ psi} > 15 \text{ psi} \text{ ok}$

REDUCE WALL THICKNESS TO 0.18" (SCH 10)

ASSUME t = 0.18" $R_i = [12.75" - 2(0.18")]/2 = 6.195"$

Do /t = 12.75/0.18 = 70.8

A = 0.00027

3(Do/t) 3 (70.8)

CALCULATE INTERNAL PRESSURE RATING FOR 0.18" WALL THK

P= SEt = 6,000 (1) (0.18) = 170 psi $\left(\frac{12.75}{2} - 0.18\right) + 0.6 (0.18)$

CHECK LIMITS: t = 0.18" < 6.195/2=3.0975"= Ri OK P=170 < 0.385 SE = 0.385 (6,000)(1) = 2,310 OK CIRCUMFERENTIAL STRESS LIMITS P.

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CHECK COLUMN STABILITY FOR AXIAL COMPRESSION ONLY:

$$S_{y} = 35,000 \text{ psi}$$

$$C_{c} = \begin{cases} 2\pi^{2}E \\ 5y \end{cases} = \begin{cases} 2\pi^{2}(10\times10^{6}) \\ 35,000 \end{cases} = 75$$

$$\frac{2L}{R} = \frac{2(420^{9})}{(12.75\%)} = 132$$

$$\frac{2L}{R} = 132 \Rightarrow C_{c} = 75$$

:
$$Fa = \frac{12\pi^2 E}{23(2Lv/R)^2} = \frac{12\pi r^2(10\times10^6)}{23(132)^2} = 2,955 psi$$

$$Sa = \frac{15}{4} = \frac{15}{16} + \left(\frac{12.75}{4} \right)^{2} + \frac{270}{4} = 270 \text{ psi}$$

$$= \frac{15}{4} \left[(2.75)^{2} - (12.75 - 2\{0.18\})^{2} \right]$$

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CALCULATE THE CRITICAL COLLAPSING PRESSURE FOR THE CHIMNEY ...

- REFR: 1) THE NEW-TYPE CODE CHART FOR THE DESIGN OF VESSELS UNDER EXTERNAL PRESSURE,
 BY: E.O. BERGMAN
 ASME TRANSACTIONS, 1952
 - 2) VESSELS UNDER EXTERNAL PRESSURE BY: D.F. WINDENBURG MECHANICAL ENGINEERING, 1937
 - 3) COLLAPSE BY INSTABILITY OF THIN CYLINDRICAL SHELLS UNDER EXTERNAL PRESSURE
 BY: D.F. WINDENBURG & C. TRILLING
 ASME TRANSACTIONS, 1934

ASME SECTION VIIL DIVI USES THE FOLLOWING COLLARSE
PRESSURE FORMULAS IN THEIR METHOD FOR CALCULATING
THE THICKNESS OF SHELLS AND TUBES UNDER EXTERNAL
PRESSURE (REFR I):

FOR INSTABILITY FAILURE BELOW THE CRITICAL LENGTH -

$$P_{CB} = \frac{2.42E}{(1-\mu^2)^{3/4}} \frac{(t/D)^{2.5}}{(4D) - 0.45(t/D)^{0.5}}$$

FOR M = 0.3 (M IS POISSON'S RATIO)

FOR 5083-0 ALUMINUM AT THE DESIGN CONDITIONS OF THIS CHIMNEY, M = 0.334 . WITH M = 0.334 INSTEAD OF 0.3, THE CONSTANT IN THE ABOVE EQUATION IS 2.64 INSTEAD OF 2.6. THIS IS A NEGLIGIBLE DIFFERENCE.

THE ABOVE FORMULA APPEARS AS THE COLLAPSE PRESSURE FORMULA IN CGA-341-1987, PG 6.

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FOR INSTABILITY FAILURE ABOVE THE CRITICAL LENGTH -

$$Pc_A = \frac{2.2 E}{1.27} \left(\frac{\pm}{D}\right)^3$$
 FOR $u = 0.3 = \frac{4}{5} \frac{1}{5} \times 0.0023$

FOR THIS CHIMNEY, t/D = 0.18/12.75 = 0.0141 < 0.023 OK THE CRITICAL LENGTH, LC, CAN BE DETERMINED BY SETTING PCB = PCA.

$$\frac{2.6 \, \text{Å} \, (\text{t/D})^{2.5}}{\left[(\frac{1-5}{0}) - 0.45 \, (\text{t/D})^{0.5} \right]} = \frac{2.2 \, \text{Å}}{1.27} \left(\frac{\text{t}}{\text{p}} \right)^{3.5}$$

$$\frac{2.6(1.27)}{2.2(t/p)^{0.5}} + 0.45(t/p)^{0.5} = -4$$

$$\frac{2.6(1.27)}{2.2(0.18/2.75)^{0.5}} + 0.45 - \left(\frac{0.18}{12.75}\right)^{0.5} = \frac{14}{12.75} = 162''$$

CHIMNEY LENGTH = 420" > 162"

FOLLOWING CGA-341-1987 PARA 3.6.2, THE MINIMUM COLLAPSING PRESSURE OF 48.74 PSI DIFFERENTIAL IS EQUIVALENT TO A 24.37 PSI DIFFERENTIAL PRESSURE WITH A SAFETY FACTOR OF Z.

THE ALLOWABLE EXTERNAL PRESSURE CALCULATED BY
THE RULES OF ASME SECTION VIII DIVION PAGE 9 OF
THESE CALCULATIONS IS 25.4 PSI.

25.4 = 24.37 ⇒ ASME SECTION VIII DIN I HAS A SAFETY FACTOR OF 2 ON COLLAPSE PRESSURE.

THIS CHIMNEY HAS A COLLAPSE PRESSURE SAFETY FACTOR OF 48.74 = 3.25.



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DETERMINE THE CIRCUMFERENTIAL STRESS AT THE COLLAPSE PRESSURE:

$$P = \underbrace{SEt}_{R+0.6t} \implies S = \underbrace{P(R+0.6t)}_{Et}$$

SECTION PROJECT SERIAL - CATEGORY SDC 4,4 ENGINEERING NOTE AMS LAYOUT CROSS SECTION FLOW AREA BLOCKED BY -12" SCH 10 ALUMINUM LN2 SHIELD SPACERS. t=0.18" 4 SPACERS AT 4 LOCATIONS .-TYP. /" THK MLI 6.625" O.D. ALUMINUM RADIATION SHIELD AND POSSIBLE VACUUM SHELL

LHe , LN2 AND SUPER-CONDUCTING COIL LEADS LOCATED IN THIS AREA.

NOTE: THE MLI SHALL BE SECURED SO THE VENTING GAS WILL NOT RIP IT OFF THE LN2 SHIELD. FOR EXAMPLE, WRAP THE MLI WITH A FABRIC, I.E. NYLON,

Box.

THIS AREA IS BLOCKED AT TEE

TO CHIMNEY BAYONET

COVER SHEET AND SECURE THE FABRIC WITH NYLON OR STEEL (CHICKEN WIRE TYPE) MESH, OR

ENCLOSE THE MLI WITH A THIN ALUMINUM JACKET.

PROJECT SECTION SERIAL - CATEGORY FERMILAB SDC 4,4 ENGINEERING NOTE SUBJECT AMS REVISION DATE PLAN CHIMNEY LENGTH THROUGH THE ZETECTOR = 11,315 - 2200 = 10,215 mm = 33.5 ftASSUME THE CHIMNEY EXTENDS 6" PAST THE DETECTOR BEFORE THE TEE IS ATTACHED. : STRAIGHT PIPE LENGTH = 33.5'+0.5' 34 SKETCH: FLANGE FOR PARALLEL PLATE OR OTHER RELIEF DEVICE MOUNTS HERE 12"x 12"x6" ALUMINUM TEE FOR TURBOMOLECULAR 6"\$ LONG RADIUS 900 ALUMINUM ELBOW-6" HIGH VACUUM RADIATION BAFFLEZ GATE VALVE AND 6" TURBOMOLECULAR 30"

> LN2 SHIELD

> > TO

CHIMNE

BAYONET

BOX

CAN BE INSTALLED ALONG
THE LENGTH OF THE LN2
SHIELD TO INCREASE HIGH
VACUUM CONDUCTANCE OUT
OF THE SHIELD. 134' TO VAC. VSL. 2

PUMP INSTALL HERE

NOTE: RADIATION BAFFLES

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